

www.e-rara.ch

**Miscellaneous Tracts on some curious, and very interesting Subjects in
Mechanics, Physical-Astronomy, and Speculative Mathematics ...**

Simpson, Thomas

London, 1757

ETH-Bibliothek Zürich

Shelf Mark: Rar 5521

Persistent Link: <https://doi.org/10.3931/e-rara-9162>

A very exact method for finding the place of a planet in its orbit, from a correction of ward hypothesis, by means of one, or more equations, applied to the motion about the upper focus.

www.e-rara.ch

Die Plattform e-rara.ch macht die in Schweizer Bibliotheken vorhandenen Drucke online verfügbar. Das Spektrum reicht von Büchern über Karten bis zu illustrierten Materialien – von den Anfängen des Buchdrucks bis ins 20. Jahrhundert.

e-rara.ch provides online access to rare books available in Swiss libraries. The holdings extend from books and maps to illustrated material – from the beginnings of printing to the 20th century.

e-rara.ch met en ligne des reproductions numériques d'imprimés conservés dans les bibliothèques de Suisse. L'éventail va des livres aux documents iconographiques en passant par les cartes – des débuts de l'imprimerie jusqu'au 20e siècle.

e-rara.ch mette a disposizione in rete le edizioni antiche conservate nelle biblioteche svizzere. La collezione comprende libri, carte geografiche e materiale illustrato che risalgono agli inizi della tipografia fino ad arrivare al XX secolo.

Nutzungsbedingungen Dieses Digitalisat kann kostenfrei heruntergeladen werden. Die Lizenzierungsart und die Nutzungsbedingungen sind individuell zu jedem Dokument in den Titelnformationen angegeben. Für weitere Informationen siehe auch [Link]

Terms of Use This digital copy can be downloaded free of charge. The type of licensing and the terms of use are indicated in the title information for each document individually. For further information please refer to the terms of use on [Link]

Conditions d'utilisation Ce document numérique peut être téléchargé gratuitement. Son statut juridique et ses conditions d'utilisation sont précisés dans sa notice détaillée. Pour de plus amples informations, voir [Link]

Condizioni di utilizzo Questo documento può essere scaricato gratuitamente. Il tipo di licenza e le condizioni di utilizzo sono indicate nella notizia bibliografica del singolo documento. Per ulteriori informazioni vedi anche [Link]



A very exact METHOD for finding the PLACE of a PLANET in its orbit, from a Correction of WARD's *hypothesis*, by means of One, or more Equations, applied to the motion about the upper focus.

Fig. 18.



ET ABPC be the ellipses in which the planet revolves about the sun in the lower focus S; let F be the upper focus, and M and *m* any two places of the planet indefinitely near to each other; and let FM, F*m*, SM, S*m* be drawn, as likewise MN, perpendicular to the greater axis AP: from the center F, with the radius FD = 1, let the circumference of a circle DE*e* be described, and from its intersection with FM draw EH perpendicular to AP: put AO (= OP) = *a*, OB (= OC) = *b*, OF (= OS) = *c*, FM = *u*, FH = *x*, EH = *y*, DE = *z*, and E*e* = *ż*: then SM being (= AP — FM) = 2*a* — *u*, by the nature of the ellipses, and FN (= FH × $\frac{FM}{FE}$) = *xu*, we have, by a known property of triangles, $\overline{SM + FM} \times \overline{SM - FM} (2a \times 2a - 2u) = 2SF \times ON (4c \times c + xu)$: from which equation *u* (FM) is found = $\frac{a^2 - c^2}{a + cx} = \frac{bb}{a + cx}$ (because *bb* = *aa* — *cc*): whence SM (= 2*a* — *u*) is also had in terms of *x*, being = $\frac{aa + cc + 2acx}{a + cx}$.

Now the area EF*e* being expressed by $\frac{1}{2}\dot{z} (= FE \times \frac{1}{2}Ee)$, we have $FE^2 (1) : FM^2 :: \frac{1}{2}\dot{z} : \text{the area } MFm = \frac{1}{2}\dot{z} \times \overline{FM}^2$. Therefore, the angle SM*m* being equal to FMA, or F*m*A (by the nature of the curve) we also have FM × M*m* : S*m* × *m*M (or FM : SM) :: the area ($\frac{1}{2}\dot{z} \times \overline{FM}^2$) of the triangle FM*m* : the area of the triangle SM*m* = $\frac{1}{2}\dot{z} \times FM \times SM$ (*Elem.* 23. 6.)

$$= \frac{\frac{1}{2}b^2\dot{z} \times aa + cc + 2acx}{(a + cx)^2} = \frac{1}{2}b^2\dot{z} + \frac{\frac{1}{2}b^2c^2\dot{z} \times 1 - xx}{(a + cx)^2} = \frac{1}{2}b^2\dot{z} + \frac{\frac{1}{2}b^2c^2y^2\dot{z}}{(a + cx)^2} = \text{the fluxion of the area ASM.}$$

fluent

fluent hereof, let $\frac{1}{a+cx}$ be resolved into the series $\frac{1}{a^2} - \frac{2cx}{a^3}$, &c. (whereof the two first terms will, here, be sufficient) by which means our fluxion will become $\frac{1}{2}b^2\dot{z} + \frac{\frac{1}{2}b^2c^2y^2\dot{z}}{a^2} - \frac{b^2c^3x\dot{z}y^2}{a^3}$; which, because $y\dot{z} = -\dot{x}$, and $x\dot{z} = y$, will be reduced to $\frac{1}{2}b^2\dot{z} + \frac{b^2c^2}{2a^2} \times -\dot{x}y - \frac{b^2c^3y^2\dot{y}}{a^3}$; whose fluent will therefore be truly expressed by $\frac{1}{2}b^2z + \frac{b^2c^2}{2a^2} \times \text{area DHE} - \frac{b^2c^3y^2}{3a^3}$, or $\frac{1}{2}b^2z + \frac{b^2c^2}{2a^2} \times \frac{1}{2}z - \frac{1}{2}xy - \frac{b^2c^3y^3}{3a^3}$ (because the area DHE = $\frac{1}{2}$ DE $\times$ DF = $\frac{1}{2}$ HF $\times$ EH = $\frac{1}{2}z - \frac{1}{2}xy$). This expression, when M coincides with P, and z is = the semi-circumference DEK (= p), will become $\frac{b^2p}{2} + \frac{b^2c^2p}{4aa} = \frac{b^2dp}{2}$ (supposing $d = 1 + \frac{cc}{2aa}$) = the area of the semi-ellipsis ABP. Therefore we have, as $\frac{b^2dp}{2} : \frac{b^2dz}{2} = \frac{b^2c^2xy}{4aa} - \frac{b^2c^3y^3}{3a^3}$ (= the area ASM) :: p (= the length of an arch of 180°) : $z - \frac{c^2xy}{2a^2d} - \frac{2c^2y^3}{3a^3d}$ = the length of an arch (A) expressing the planet's mean anomaly : from which equation, $z = A + \frac{c^2xy}{2a^2d} + \frac{2c^2y^3}{3a^3d} = A + \frac{c^2}{4a^2d} \times \overline{\text{fin. } 2z} + \frac{2c^3}{3a^3d} \times \overline{\text{fin. } z}^3$ (because xy , or co-fin. $z \times \text{fin. } z = \frac{1}{2} \text{fin. } 2z$): where the two last terms being very small in comparison of the others (and, therefore, z nearly = A), we may, instead of fin. z and fin. $2z$, substitute fin. A and fin. 2A; by which means we have $z = A + \frac{cc}{4a^2d} \times \text{fin. } 2A + \frac{2c^3}{3a^3d} \times \overline{\text{fin. } A}^3$. From whence it appears, that, in order to have the angle AFM at the upper focus, the mean anomaly (A) of the planet at the time given, must be increased by the quantity, or correction $\frac{cc}{4a^2d} \times \text{fin. } 2A + \frac{2c^3}{3a^3d} \times \overline{\text{fin. } A}^3$.

But to express the value of this correction in seconds of a degree (which in practice is the most commodious) it will

will be, as 3,1416 (the length of an arch of 180 degrees) is to 648000 (the number of seconds in that arch) so is $\frac{cc}{4a^2d} \times \text{fin. } 2A + \frac{2c^3}{3a^3d} \times \overline{\text{fin. } A}^3$ to $51567 \times \frac{c^2}{a^2d} \times \text{fin. } 2A + 137513 \times \frac{c^3}{a^3d} \times \overline{\text{fin. } A}^3 =$ the number of seconds in the said correction: the logarithm of the latter term of which will, therefore, be $= 5,1383 - \log. d + 3 \log. \frac{c}{a} + 3 \log. \text{fin. } A$; and that of the former $= 4,7123 - \log. d + 2 \log. \frac{c}{a} + \log. \text{fin. } 2A$. But, to render these expressions still more convenient for practice, the log. of d , by reason of its smallness, may be, either, intirely neglected, or else so assumed, to be nearly a mean of what it is known to be in the planetary orbits. Accordingly, by assuming the excentricity $c = \frac{8}{100}$ of the mean distance (which is a small matter less than the excentricity of *Mars*, but something greater than *those* of the *Moon*, *Saturn*, and *Jupiter*), the value of $d (= 1 + \frac{cc}{2aa})$ will be $= 1,0032$, and its logarithm $= 0,0014$. Whence the log. of the former part of our correction, by substituting this value, will be $5,1369 + 3 \log. \frac{c}{a} + 3 \log. \text{fin. } A = 3 \times 1,7123 + \log. \frac{c}{a} + \log. \text{f. } A$; and *that* of the latter $= 4,7109 + 2 \log. \frac{c}{a} + \log. \text{fin. } 2A$: which, expressed in words at length, give the following

Practical Rules.

1°. To the sum of the constant logarithm 1,7123 and the log. of the excentricity in parts of the mean distance, add the log. sine of the mean anomaly; the sum (rejecting the radius) being tripled, will give the log. of the first equation (in seconds) to be added to the mean anomaly.

2°. To the sum of the constant log. 4,7109 and twice the log. of the excentricity, add the log. sine of twice the mean anomaly; the sum (rejecting the radius) will be the log. of the second equation; to be added or subtracted, according as the mean anomaly is less, or greater than 90 degrees. Here

Here, and in what follows, the anomaly is to be always reckoned from, or to the aphelion, the nearest way; in which the seconds may be omitted, in computing the proposed corrections. Which corrections being made, the true anomaly, or angle at the lower focus, will be had from the common proportion, by saying, as the aphelion-distance is to the perihelion-distance, so is the tangent of half the mean anomaly, thus corrected, to the tangent of half the true anomaly sought.

As an example of the use of the method here laid down, let the excentricity be supposed = 0,048219 (being that assigned, by Dr. HALLEY, to the orbit of Jupiter) and let the mean anomaly be 45° .

Then, log. excent. . .	2,6832	2 log. excent. . . .	3,3664
+ const. log. . .	1,7123	+ const. log. . . .	4,7109
= log. for 1st equ. .	0,3955	= log. for 2d equ. .	2,0773
log. fin. anom. . .	9,8494	log. fin. 2 anom. .	10,0000
	0,2449	2d equ. = $119^{\circ}\frac{1}{2}$	2,0773
first equ. = $5^{\circ}\frac{1}{2}$	0,7347		

From 1,978537 = log. of perihelion-dist. 0,951781,
 subtr. 0,020452 = log. of the aphelion-dist. 1,048219;
 the rem. 1,958085, will be a (3d) const. log. for this orbit: to which
 add 9,617596 = log. tang. $\frac{1}{2}$ cor. anom. $22^{\circ} 31' 2\frac{1}{2}''$,
 so shall 9,575681 = log. tang. $\frac{1}{2}$ true anomaly $20^{\circ} 37' 40''$:
 whose double, $41^{\circ} 15' 20''$, is therefore the true anomaly required.

The same excentricity being retained; let the mean anomaly be, now, supposed 120 degrees.

Then, log. fin. anom. .	9,9375	log. fin. 2 anom. . .	9,9375
log. for first equat. .	0,3955	log. for 2d equation .	2,0773
	0,3330	2d equat. = $103^{\circ}\frac{1}{2}$	2,0148
first equat. = $10''$	0,9990		

Hence $120^{\circ} + 10'' - 1^{\circ} 43\frac{1}{2}'' = 119^{\circ} 58' 26\frac{1}{2}'' =$ the cor. anomaly.

Therf. log. tang. $\frac{1}{2}$ cor. anom. $59^{\circ} 59' 13\frac{1}{2}''$. . . 10,238331
 + third const. log. for this orbit 1,958085
 = log. tang. $\frac{1}{2}$ true anom. $57^{\circ} 32' 10''$ 10,196416
H
Whence

Whence the anomaly itself is given $= 3^{\circ} 25' 4'' 20'''$.

If the excentricity be assumed $= 0,066777$ (being the greatest that Dr. HALLEY gives to the lunar orbit) the two constant logarithms, to be added to the sines of the mean anomaly and of its double, will be 0,5369 and 2,3601: whence if the said anomaly be taken $= 50^{\circ}$ (at which, according to the *Doctor's Table, pro expediendo calculo æquationis centri lunæ*, the whole correction is a *maximum*), the former part of the said correction will be found $= 18\frac{1}{2}''$, and the latter part $= 225\frac{1}{2}''$: therefore the sum of both is $244''$, or $4' 4''$; agreeing, exactly, with the quantity given in the *Table*. And in the very same manner, the proper corrections corresponding to other anomalies and excentricities may be computed; the error never amounting to above a single second in any of the planets, except *Mars* and *Mercury*: in the place of *Mars*, the greatest error will be two or three seconds; and in that of *Mercury*, about as many minutes.—As to the *Earth* and *Venus*, the second equation, alone, will be sufficient to give their places to less than a second.

To obtain a farther correction, which will be necessary when the orbit is very eccentric, we may (instead of the two first only) make use of a greater number of terms of the series $\frac{1}{a^2} - \frac{2cx}{a^3} + \frac{3c^2x^2}{a^4} - \frac{4c^3x^3}{a^5} \&c. (= \frac{1}{a+cx})$; by which means the fluxion $(\frac{1}{2}b^2\dot{z} + \frac{\frac{1}{2}b^2c^2y^2\dot{z}}{a+cx})$ of the area MSA described about the focus S, which is proportional to the time, will be here represented by

$\frac{1}{2}b^2\dot{z} + \frac{1}{2}b^2 \times e^2 y^2 \dot{z} - 2e^3 y^2 x \dot{z} + 3e^4 y^2 x^2 \dot{z} - 4e^5 y^2 x^3 \dot{z} \&c. (supposing e = \frac{c}{a})$. But it is well known that $y^2 (\sin. z)^2 = \frac{1}{2} - \frac{1}{2} \cosin. 2z$; whence $y^2 x (= y^2 \times \cosin. z) = \frac{1}{2} \cosin. z - \frac{1}{2} \cosin. 2z \times \cosin. z = \frac{1}{2} \cosin. z - \frac{1}{4} \cosin. z - \frac{1}{4} \cosin. 3z^* = \frac{1}{4} \cosin. z - \frac{1}{4} \cosin. 3z$; and therefore $y^2 x^2$

* This, and all that follows to the same purpose, is nothing more than the application of the THEOREM, That the rectangle of the co-sines of any two angles (the radius being unity) is equal to half the sum of the co-sines of the sum and difference of those angles.

$= \frac{1}{4} \text{co-fin. } z \times \text{co-fin. } z - \frac{1}{4} \text{co-fin. } 3z \times \text{co-fin. } z = \frac{1}{8} + \frac{1}{8} \text{co-fin. } 2z - \frac{1}{8} \text{co-fin. } 2z - \frac{1}{8} \text{co-fin. } 4z = \frac{1}{8} - \frac{1}{8} \text{co-fin. } 4z$: whence also $y^2 x^3 = \frac{1}{8} \text{co-fin. } z - \frac{1}{8} \text{co-fin. } 4z \times \text{co-fin. } z = \frac{1}{8} \text{co-fin. } z - \frac{1}{16} \text{co-fin. } 3z - \frac{1}{16} \text{co-fin. } 5z$. Which values being now substituted for their equals, our fluxion, above given, will become

$$\frac{1}{2} b^2 \dot{z} + \frac{1}{2} b^2 \text{ into } e^2 \times \frac{1}{2} \dot{z} - \frac{1}{4} \times 2 \dot{z} \text{co-fin. } 2z - 2e^3 \times \frac{1}{4} \dot{z} \text{co-f. } z - \frac{1}{12} \times 3 \dot{z} \text{co-f. } 3z + 3e^4 \times \frac{1}{8} \dot{z} - \frac{1}{12} \times 4 \dot{z} \text{co-f. } 4z - 4e^5 \times \frac{1}{8} \dot{z} \text{co-f. } z - \frac{1}{48} \times 3 \dot{z} \text{co-f. } 3z - \frac{1}{80} \times 5 \dot{z} \text{co-f. } 5z + \text{Ec.}$$

and, consequently, the fluent thereof $= \frac{1}{2} b^2 z + \frac{1}{2} b^2$ into $e^2 \times \frac{1}{2} z - \frac{1}{4} \text{fin. } 2z - 2e^3 \times \frac{1}{4} \text{f. } z - \frac{1}{12} \text{f. } 3z + 3e^4 \times \frac{1}{8} z - \frac{1}{12} \text{fin. } 4z - 4e^5 \times \frac{1}{8} \text{fine } z - \frac{1}{48} \text{fine } 3z - \frac{1}{80} \text{fine } 5z \text{ Ec.} = \frac{1}{2} b^2$ into $1 + \frac{1}{2} e^2 + \frac{1}{8} e^4 \times z - \frac{1}{2} e^3 + \frac{1}{2} e^5 \times \text{fin. } z - \frac{1}{4} e^2 \text{fin. } 2z + \frac{1}{6} e^3 + \frac{1}{12} e^5 \times \text{fin. } 3z - \frac{3e^4}{32} \text{fin. } 4z + \frac{e^5}{20} \text{fin. } 5z$ (supposing all

such terms wherein e rises to more than five dimensions, to be disregarded). This expression, when $z = p$ (= the semi-circumference AEK) becomes $\frac{1}{2} b^2 \times 1 + \frac{1}{2} e^2 + \frac{1}{8} e^4 \text{ Ec.} \times p = \frac{1}{2} b^2 p \times \sqrt{1 - ee}^{-\frac{1}{2}} = \frac{\frac{1}{2} b^2 p}{\sqrt{1 - \frac{cc}{aa}}} = \frac{\frac{1}{2} b^2 p \times a}{\sqrt{a^2 - c^2}} = \frac{1}{2} abp$ = the area

of the femi-ellipse ABP. Hence it will be, as $\frac{1}{2} abp$ (area ABP) : $\frac{1}{2} abz - \frac{1}{2} b^2 \times \frac{1}{2} e^3 + \frac{1}{2} e^5 \times \text{fin. } z - \frac{1}{2} b^2 \times \frac{1}{4} e^2 \text{fin. } 2z \text{ Ec.}$ (area ASM) :: p (the length of an arch of 180°) : $z - \frac{b}{a} \times \frac{1}{2} e^3 + \frac{1}{2} e^5 \times \text{fin. } z - \frac{b}{a} \times \frac{e^2}{4} \text{fin. } 2z \text{ Ec.} =$ the length of the arch A , expressing the planet's mean anomaly: whence, by substituting $f = \frac{b}{a}$, we have $A = z - \frac{1}{1 + e^2 \times \frac{1}{2} f e^3 \text{fin. } z - \frac{1}{4} f e^2 \text{fin. } 2z + 1 + \frac{1}{2} e^2 \times \frac{1}{6} f e^3 \text{fin. } 3z - \frac{3 f e^4}{32} \times \text{fin. } 4z + \frac{f e^5}{20} \times \text{fine } 5z$.

Now, to find from hence the value of z , in terms of A and the fines of its multiples, we may, for a first approximation, assume $1 + e^2 \times \frac{1}{2} f e^3 \text{fin. } z + \frac{1}{4} f e^2 \text{fin. } 2z \text{ Ec.}$ as equal to $1 + e^2 \times \frac{1}{2} f e^3 \text{fin. } A + \frac{1}{4} f e^2 \text{fin. } 2A - 1 + \frac{1}{2} e^2 \times \frac{1}{6} f e^3 \text{fin. } 3A \text{ Ec.}$ which last quantity being denoted by Q , we shall have $A = z - Q$, and consequently $z = A + Q$.

But $\text{fin. } z (= \text{fin. } A + Q) = \text{fin. } A \times \text{co-fin. } Q + \text{fin. } Q \times \text{co-fin. } A = \text{fin. } A + Q \times \text{co-fin. } A$, nearly (because Q being very small, its co-sine will differ insensibly from the radius, and the sine very little from the arch itself). In the same manner, $f. 2z (= f. 2A + 2Q) = f. 2A + 2Q \times \text{co-f. } 2A$, $\text{fin. } 3z = f. 3A + 3Q \times \text{co-f. } 3A$, &c. Which values being therefore substituted in the given equation, it will become $A = z - Q - Q \times \frac{1 + e^2 \times \frac{1}{2} f e^3 \text{co-f. } A + \frac{1}{2} f e^2 \text{co-f. } 2A - 1 + \frac{1}{2} e^2 \times \frac{1}{2} f e^3 \text{co-f. } 3A}{1 + e^2 \times \frac{1}{2} f e^3 \text{co-fin. } A} \&c.$ But the first term of the series, $1 + e^2 \times \frac{1}{2} f e^3 \text{co-fin. } A$, drawn into $-Q$ (if all terms having more than five dimensions of e be neglected) will be barely $= \frac{1}{2} f e^3 \times \text{co-fin. } A \times -\frac{1}{4} f e^2 \text{fin. } 2A = -\frac{1}{8} f^2 e^5 \times \text{co-fin. } A \times \text{fin. } 2A = -\frac{1}{16} f^2 e^5 \times \text{fin. } 3A + \text{fin. } A$.

In the same manner $\frac{1}{2} f e^2 \text{co-fin. } 2A \times -Q = -\frac{1}{2} f e^2 \text{co-fin. } 2A \times \frac{1}{2} f e^3 \text{fin. } A + \frac{1}{4} f e^2 \text{fin. } 2A - \frac{1}{6} f e^3 \text{fin. } 3A = -\frac{1}{8} f^2 e^5 \times \text{fin. } 3A - \text{fin. } A - \frac{1}{16} f^2 e^4 \times \text{fin. } 4A + \frac{1}{24} f^2 e^5 \times \text{fin. } 5A + \text{fin. } A$: And, lastly, $-1 + \frac{1}{2} e^2 \times \frac{1}{2} f e^3 \text{co-fin. } 3A \times -Q$ equal to $\frac{1}{2} f e^3 \text{co-fin. } 3A \times Q = \frac{1}{2} f e^3 \times \text{co-fin. } 3A \times \frac{1}{4} f e^2 \text{fin. } 2A = \frac{1}{16} f^2 e^5 \times \text{fin. } 5A - \text{fin. } A$. The sum of all which will be $\frac{1}{24} f^2 e^5 \text{fin. } A - \frac{1}{16} f^2 e^5 \text{fin. } 3A - \frac{1}{16} f^2 e^4 \text{fin. } 4A + \frac{5}{24} f^2 e^5 \text{fin. } 5A$; which added to $z - Q$ (or its equal, $z - 1 + e^2 \times \frac{1}{2} f e^3 \text{fin. } A - \frac{1}{4} f e^2 \text{fin. } 2A$, &c.) gives $A = z - 1 + e^2 - \frac{1}{2} f e^2 \times \frac{1}{2} e^3 f \text{fin. } A - \frac{1}{4} e^2 f \text{fin. } 2A + 1 + \frac{1}{2} e^2 - \frac{3}{8} e^2 f \times \frac{1}{6} e^3 f \text{fin. } 3A - \frac{3f + 2f^2}{32} \times \frac{e^4}{\text{fin. } 4A} + \frac{1}{20} f + \frac{5}{8} f^2 e^5 \times \text{fin. } 5A$. From whence

we have $z = A + 1 + \frac{11ee}{12} \times \frac{1}{2} e^3 f \text{fin. } A + \frac{1}{4} e^2 f \text{fin. } 2A - 1 - \frac{5ee}{8} \times \frac{1}{6} e^3 f \text{fin. } 3A + \frac{5e^4 f^2}{32} \text{fin. } 4A - \frac{37e^5 f}{240} \text{fin. } 5A$, very near;

because in all terms having more than three dimensions of e , the quantities 1 and f may be used indifferently, for each other, without producing any errors but such as consist of more than five dimensions of the converging quantity e .

But, since it is known that $\text{fin. } 3A = 3S - S^3$, and $\text{fin. } 5A = 5S - 20S^3 + 16S^5$ (S being the sine of A) we shall, by substituting these values in the 2d, 4th, and 6th terms (after proper

proper reduction) have $z = A + 1 + 4ee \times \frac{2}{3}e^3fS^3 - \frac{37e^5f}{15}S^5$

$+ \frac{1}{4}e^2f \times \text{fin. } 2A + \frac{5e^4f^2}{32} \times \text{fin. } 4A$. Hence it appears that

$$\frac{180 \times 60 \times 60}{3,1416} \times 1 + 4ee \times \frac{2}{3}e^3fS^3 - \frac{37e^5f}{15}S^5 + \frac{1}{4}e^2f \text{ fin. } 2A + \frac{5e^4f^2}{32} \text{ fin. } 4A$$

will express the number of seconds, to be added to the mean anomaly, in order to have the angle AFM at the upper focus of the ellipsis, corresponding to that anomaly. From whence is deduced the following method of calculation.

Let F denote the logarithm of half the lesser axis divided by half the greater, E the log. of the eccentricity divided by half the greater axis, G the log. of the sum of the squares of half the greater axis and twice the eccentricity divided by the square of half the greater axis.

$$\text{Take } P = 1,71277 + E + \frac{1}{3}F + \frac{1}{3}G,$$

$$Q = 1,14130 + E + \frac{1}{5}F,$$

$$R = 4,71236 + 2E + F,$$

$$S = 4,50824 + 4E + 2F;$$

then the logarithms of four equations (in seconds), to be applied to the mean anomaly (A), will be

$$3 \times P + \log. \text{fin. } A - \log. \text{rad.}$$

$$5 \times Q + \log. \text{fin. } A - \log. \text{rad.}$$

$$R + \log. \text{fin. } 2A - \log. \text{rad.}$$

$$S + \log. \text{fin. } 4A - \log. \text{rad.}$$

Of which equations the first is always to be added, and the second always subtracted; the other two being to be added, or subtracted, according as the sines of their respective arguments, $2A$ and $4A$, are positive, or negative.

The two principal of these equations agree with those before given; and are the same, in effect, with the two equations laid down (without demonstration) by Sir ISAAC NEWTON, in the Scholium to the 3rd Proposition of the first book of his *Principia*. The latter of which, in the *Laws of the Moon's Motion*, prefixed to that Work, seems to be represented, as defective; it being there asserted, that, *the inequality in the motion of a planet about the upper focus, consists of three parts*; as if the nature of the subject admitted of just that number, and no more

more; whereas the parts, or equations arising in the consideration, are without number, being the terms of an infinite series, wherein the eccentricity is the converging quantity. Sir ISAAC NEWTON has given two terms of this series, which are right; but the new equation added by his *Commentator*, is not so; the sign thereof, the coefficient, and the law by which it increases and decreases, being all different from what they ought to be.

This equation (expressed according to the above notation, where 1 represents half the greater axis, f half the lesser axis, e the eccentricity, and A the mean anomaly) he makes to be $-\frac{1}{2}e^4 \times \sin. A^3 \times \text{co-sin. } A$. But it ought to be $+\frac{3f+2f^2}{32} \times e^4 \sin. 4A$ ($=\frac{5e^4}{32} \times \sin. 4A$, nearly), as is shewn above; this being the 3d term of the general series, and the next in order after those given by Sir ISAAC NEWTON; who appears, more than once*, to have been disadvantageously (I might say, unfairly) represented, and that, under the covers of his own book: a circumstance that cannot be attended to without some concern and dislike, by those who entertain a due regard for the merit of an Author to whom the mathematical world is so much indebted.

I shall now put down one single example of the use of the equations above derived; wherein I shall suppose the eccentricity to be $\frac{2.00589}{1.0000000}$ parts of the semi-transverse axis (the same as is assigned by Dr. HALLEY to the orbit of Mercury). Here, then, we have $E = -1.313635$; $F (= \log. \sqrt{1-ee} = \frac{1}{2} \log. 1-ee) = -0.009406$; $G (= \log. 1+4ee) = 0.068024$; whence $P = 1.046$; $Q = 0.453$; $R = 3.3302$; $S = 1.744$:

* In the 28th Proposition of his third book, it is found that the moon's distance from the earth in the syzgies is to the distance in the quadratures (setting aside the consideration of eccentricity) as 69 to 70; which is confirmed by what is demonstrated in a subsequent part of this our Work, as well as by the calculations of others; nevertheless the truth of this proportion is called in question, and a new one is laid down, which makes the said distances to be in the proportion of 59 to 60. See *Laws of the Moon's Motion*, p. 11 and 12.

which

which values will serve in all cases belonging to this orbit. Whence, supposing the given anomaly to be 50° , we have

1,0460	0,4530	3,3302	1,744
<u>9,8842</u>	<u>9,8842</u>	<u>9,9933</u>	<u>9,534</u>
0,9302	0,3372	3,3235	1,278.
<u>3</u>	<u>5</u>		
2,7906	1,6860		

Of which resulting logarithms, the numbers corresponding are $617\frac{1}{2}$, $48\frac{1}{2}$, $2106\frac{1}{4}$, and 19; whereof the first and third being added, and the other two subtracted, we have $50^\circ 44' 16''$ for the corrected anomaly, or the angle at the upper focus; whereof the half is $25^\circ 22' 8''$:

$$\begin{aligned} &\text{Therefore log. tang. } 25^\circ 22' 8'' \dots\dots\dots 9,6759338 \\ &+ \text{log. of the ratio of the gr. and least dist. } 1,8185730 \\ &= \text{log. tang. } \frac{1}{2} \text{ true anomaly } 17^\circ 20' 28'' \quad 9,4945068. \end{aligned}$$

Which conclusion is true to a second. Nor will the error, in any part of this orbit, amount to more than about two or three seconds.—If you would have the result depended on to a single second, or if the orbit be supposed still more eccentric than that of Mercury, then the following method may be of use.

Say, as half the greater axis of the ellipsis is to the eccentricity, so is the sine of the mean anomaly to the sine of an angle; which subtract from the mean anomaly, and to the log. sine of the remainder (which I call the eccentric anomaly) add the sum of the log. of the eccentricity and the constant log. 1,758123: the aggregate (rejecting the radius) will be the logarithm of an angle, in degrees and decimal parts; which, subtracted from the angle first found, leaves a correction to be added (under its proper sign) to the mean anomaly: with which corrected anomaly, let the whole operation be repeated, if needful, by always adding the last correction to the mean anomaly. Then it will be, as the greater semi-axis of the ellipsis is to the lesser, so is the tangent of the corrected anomaly to the tangent of the angle at the upper focus of the ellipsis: whence the angle at the lower focus, or the true anomaly, may also

also be known by the common proportion, and that to any assigned degree of exactness.

This method, in all the planets, except *Mars* and *Mercury*, answers to a second, at one operation. In the former of these two, the error, when greatest, will amount to about three or four seconds; and in the latter, to nearly as many minutes; in which case, three operations will be necessary: but in order to avoid that trouble, the following calculation may be used; which is so exact as to answer, even in the orbit of *Mercury*, to less than half a second, without repeating the operation.

Add together twice the log. of the eccentricity, the log. secant of the angle first found (as above), and the log. co-sine of the sum of the eccentric anomaly and mean anomaly once corrected (in all of which angles the seconds may be neglected). The aggregate (subtracting twice the radius) will be the log. of a fraction to be added to unity, when the said sum of anomalies is between 90° and 270° ; but otherwise, subtracted therefrom: then the log. of this sum, or remainder being subtracted from the log. of the first correction, you will have the log. of the true correction to be added (under its proper sign) to the mean anomaly given.

Thus, for example, let the mean anomaly be 70° ; and let the eccentricity be $= 0,20589$ (as in the preceding example).

$$\begin{array}{r} \text{Here, fin. mean anom. } 70^\circ \dots\dots\dots 9,9729858 \\ \text{log. eccent. } \dots\dots\dots 1,3136353 \\ \therefore \text{ ang. first found } = 11^\circ 9',33 \quad 9,2866211; \end{array}$$

whence $58^\circ 50',67 =$ the eccentric anomaly :

$$\begin{array}{r} \text{whose sine } \dots\dots\dots 9,9323552 \\ + \text{ the log. proper for the orbit } \dots\dots\dots 1,0717583 \\ = \text{ log. of } 10^\circ,0952 \dots\dots\dots 1,0041135: \end{array}$$

which angle, or its equal, $10^\circ 5',71$, being subtracted from $11^\circ 9',33$, leaves the first correction $1^\circ 3',62 = 63',62$.

Moreover,

Moreover, by adding together 70° , $1^{\circ} 3'$, and $58^{\circ} 50'$, we have

$$\begin{array}{rcl} 129^{\circ} 53', \text{ whose co-sine} & & 9,8070 \\ + \text{sec. } 11^{\circ} 9' \text{ ang. first found} & . . . & 10,0083 \\ + \text{twice log. eccent.} & & \underline{2,6272} \\ = \text{log. of } 0,0272 & & 2,4425 \end{array}$$

$$\begin{array}{rcl} \text{Log. first cor. } 63',62 & & 1,80359 \\ - \text{log. } 1,0272 & & 0,01187 \\ = \text{log. true cor. } 61',9 & . . . & \underline{1,79172} \end{array}$$

$$\begin{array}{rcl} \text{Now the mean anom. } + \text{true cor.} = 71^{\circ} 1',9, & \} & 10,4638087 \\ \text{whose log. tang.} & & \\ + \text{log. of the semi-conjugate axis} & & \underline{1,9905940} \\ = \text{log. tang. of } 70^{\circ} 38',82 & & 10,4544027 \end{array}$$

$$\begin{array}{rcl} \text{And the log. tang. of the half hereof, } 35^{\circ} 19',41 & & 9,8504350 \\ + \text{log. of the ratio of the gr. and least distances} & & \underline{1,8185730} \\ = \text{log. tang. of } 25^{\circ} 1',02 & & 9,6690080 \end{array}$$

whose double $50^{\circ} 2',04$, or $50^{\circ} 2' 2''$, is the true anomaly required.

